



Investigation on Structural Behaviour of Cold-formed Steel Built-up Columns - Modified Direct Strength Method

Sivaganesh Selvaraj¹, Mahendrakumar Madhavan²

Abstract

An investigation of the behavior of cold-formed steel (CFS) built-up I column assemblies are presented. To study the interactive buckling mode of failures, the built-up column assembly failure modes and test results from the literature are collected, totaling 813 data points. The parameters including different shapes dimensions, lengths, spacing between the fastener connections, and slendernesses are investigated. The result shows that the local buckling deformations caused the built-up shape assembly columns to fail predominantly in interactive local and flexural-torsional buckling. The effect of spacing between the fastener connections and boundary conditions is also prevalent in the failure modes. The appropriateness of the AISI's maximum intermediate connection spacing limitation is verified to prevent instability failures. The design results assessment indicated that the current AISI's DSM design curve for interactive buckling is unconservative for the CFS built-up columns with predominant interactive local and overall buckling failure mode vulnerability.

1. Introduction

Over the past several years it has been found from experimental and numerical research that the strength erosion in cold-formed steel (CFS) structural members occurs due to material, geometric non-linearity and interaction failure both in non-slender sections and slender sections (Loughlan 1979). In general, the ultimate load of the CFS members is affected by local (L), distortional (D), and overall (O) modes of failures, and yielding (Y) based on the plate and overall slenderness (λ). The CFS members are also affected by the interaction failure involving LDO modes depending on the shape geometry (unrestrained length, shape and dimensions), end support conditions, F_y/F_{cr} ratio, and $F_{cr \max}/F_{cr \min}$ ratio. The interaction failure modes reduce the ultimate load of the members significantly below its corresponding independent modes, for example, LO (local-overall buckling interaction failure) < L (pure local failure). This is because, when two critical failure modes interact the post-buckling strength, ultimate load and equilibrium paths are likely to be affected (Dinis and Camotim 2011; Schafer and Peköz 1999; Yang and Hancock 2004; Ungureanu and Dubina 2004; Dinis et al 2007; Camotim et al. 2008; Kwon et al. 2009). The critical review, assessment, classification and slenderness limits of independent and interaction failure of single

¹ Research Assistant Professor, Department of CEE, The Hong Kong Polytechnic University, Hong Kong, sivaganesh.selvaraj@polyu.edu.hk

² Professor, Department of Civil Engineering, Indian Institute of Technology Hyderabad, India, mkm@ce.iith.ac.in

CFS channel sections are available in dos Santos et al. (2012), Martins et al. (2015), Dinis et al. (2018), Young et al. (2018), Camotim et al. (2020a; 2020b), and Zhang and Alam (2023). All the above-mentioned interaction curve design procedures (Figure. 1) are for interaction failure with two large slendernesses, but there are also other factors causing significant interaction failure. The CFS built-up cross sections are expected to have larger local and overall buckling stability than single or distinct sections and are purely subjected to distinct failure modes. Most of the researchers proposed a modified best fit Direct Strength Method (DSM) design equation or resistance factors to account for these interaction failures (Zhang and Young 2012 and 2015; Lu et al. 2017; Li and Young 2022a; 2022b; Mahar et al. 2023; Selvaraj and Madhavan 2022a; 2022b; 2023a). The above approach continues without finding actual parameters or factors causing the interaction failure and incorporating them in the DSM design equations (Zhang and Alam 2023). Other than that, those authors should have used the interactive design curve presented in Figure. 1 for assessment of their results, though conservative it is better to be safe. The investigation of interaction failure modes in CFS built-up shapes columns is scarcer. Therefore, this paper focused on exploring the various factors causing interaction failure and proposing future directions for accurate ultimate load calculation of CFS built-up sections subjected to axial compression. A total of 813 test results are used in this paper.

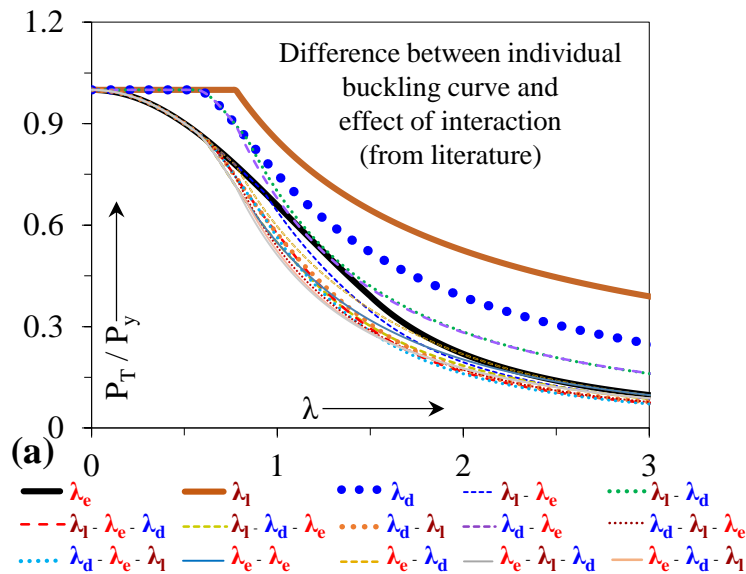


Figure 1: Interaction failure curves of the DSM method

2. Cold-formed Steel Built-up Column Results Database with Interactive Failure Modes

A cold-formed steel built-up column test results database was aggregated with a large range of parameters for analysis. The database comprises test results from seventeen literatures, including ten different built-up shapes (Figure. 2), different boundary conditions, unrestrained length of the column, local slenderness, distortional slenderness, overall buckling slenderness, Young's modulus, yield stress and spacing between the fastener connections (a), totalling to 813 test results as shown in Table 1.

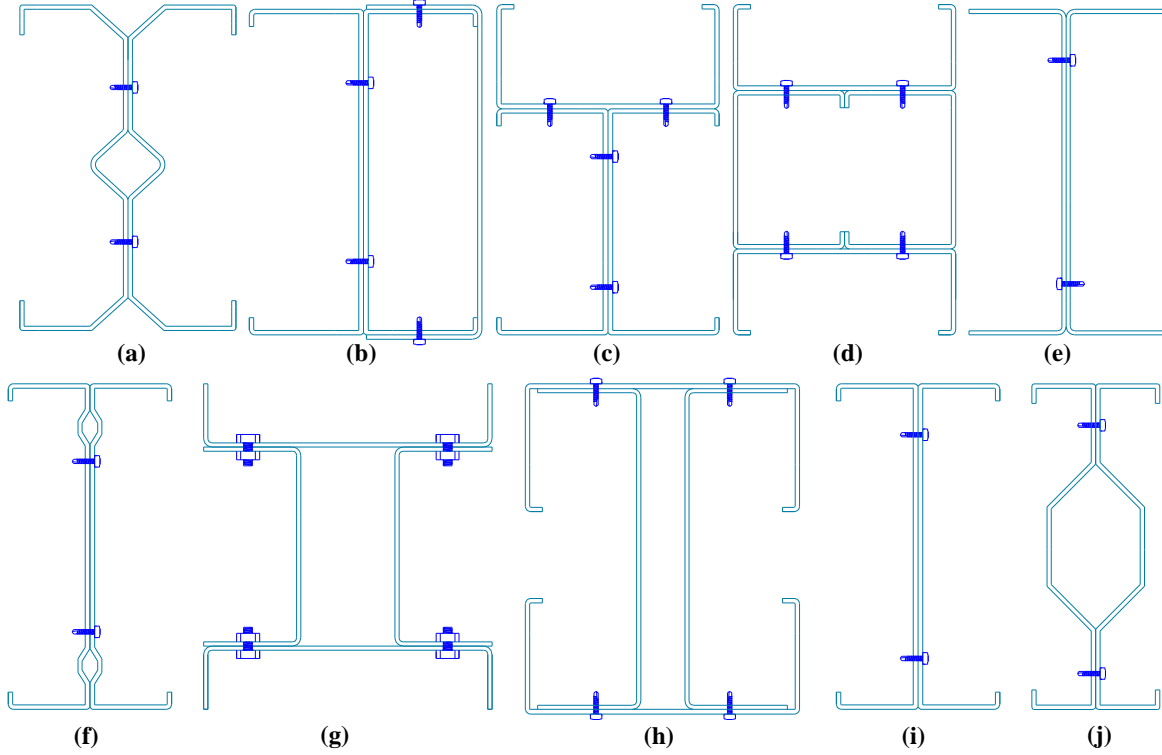


Figure 2: Various built-up shapes investigated in this study (results taken from the literature)

Table 1. Built-Up Column Test Results Database for Analyzing Interactive Buckling

Cross-section Figure (Reference)	Boundary conditions	Length (Le) (m)	Slenderness			E (GPa)	Yield Stress (MPa)	a (mm)
			Local	Distorsional	Global			
Fig. 2a - Vy et al. (2021)	$K = 0.5$	0.6-0.8	1.04- 2.92	1.20- 1.79	0.16- 0.65	205	615	30- 400
Fig. 2b-1c - Meza et al. (2020a and 2020b)	$K = 0.5$ $K = 1$	0.4-3.147	1.95- 2.75	0.17- 1.25	0.23- 1.32	206	259.5- 507.5	167- 960
Fig. 2d - Vy et al. (2021), Sang et al. (2023), Fratamico et al. (2018), Craveiro et al. (2016), Zhou et al. (2020), Lu et al. (2017)	$K = 0.5$ $K = 0.7$ $K = 1$	0.179-4.0	0.48- 2.92	0.60- 1.79	0.09- 1.48	193.9 -216	292.9- 615	30- 2000
Fig. 2e-2f - Zhang and Young (2012 and 2015) and Li and Young (2022b)	$K = 0.5$ $K = 1$	0.150-2.0	0.48- 3.59	0.25- 1.79	0.22- 2.21	204- 221	604- 697	100
Fig. 2g - Sang et al. (2022)	$K = 0.5$ $K = 1$	0.362-3.5	2.18- 2.42	0.38- 1.45	0.14- 1.55	193.9 - 196.1	283- 292.95	45- 1000
Fig. 2h-2i - Phan et al. (2021 and 2022)	$K = 0.5$ $K = 1$	0.25-6.15	1.26- 2.38	1.19- 1.75	0.14- 3.55	213.3 32	608- 628	100- 1000
Fig. 2j - Selvaraj and Madhavan (2021a and 2022a)	$K = 0.5$	0.50-0.90	1.52- 1.97	0.19- 0.19	0.28- 1.40	212.0 79	377.4	100- 900

Local $\lambda_l = (F_y/F_{crd})^{0.5}$; distortional $\lambda_d = (F_y/F_{crd})^{0.5}$; Global $\lambda_e = (F_y/F_{cre})^{0.5}$; L_e - unrestrained length of the member (overall length multiplied by K); a - spacing between fastener connections; K - effective length factor; $K = 0.5$ - Fixed end columns (F-F); $K = 0.7$ - semi-rigid conditions (S-R);

$K = 1$ - Pin ended columns (P-P); For all the results collected, the critical elastic buckling stresses were determined using the method suggested by Selvaraj and Madhavan (2019; 2021b; 2021c; 2022c; 2022d; 2023b; 2023c; 2023d; 2023e)

3. Factors Causing Interaction Failure

In general, the visible interaction failure arises when two slenderness ratios are more than unity (two P_{cr} is less than P_y), however in the present study specimens with susceptibility only to one of the distinct failure modes experienced visible interaction failure compared to its corresponding failure curve as shown in Figures. 3 and 4. Therefore, the CFS built-up column results in Table 1 are analyzed with different parameters to assess the various other reasons for interaction failure.

3.1 Interaction Failure: Due to Built-up Assembly Shapes

The test results of all the built-up columns with vulnerability only to local failure ($\lambda_l > 1$) are compared with design calculations in Figure. 3. It can be observed from Figure. 3 that most of the cross-section except Figure. 2b displayed interaction failure. It can be noted that all these built-up cross-sections are prone to fail first in flange local failure. Therefore, the non-uniform local failure might cause a shift in the center of gravity and interact with global failure. This shift in the center of gravity will induce flexural torsional failure in CFS thin-walled sections (Rasmussen and Hancock 1993; Young and Rasmussen 1999a; 1999b; Selvaraj and Madhavan 2021a and 2022a).

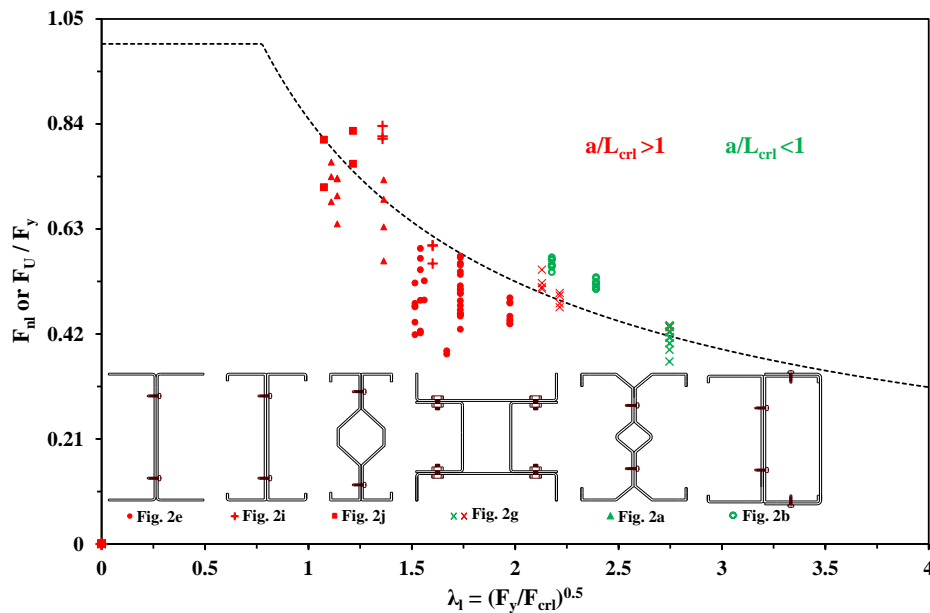


Figure 3: Interaction Failure Due to Built-up Assembly Shapes

3.2 Interaction Failure: Due to Buckling Half-wavelengths and Spacing between the Fastener Connections (a)

Studies have shown that the larger spacing between the fastener connections (a) can also cause an individual section failure separation (Selvaraj and Madhavan 2021a and 2022a) or local-overall buckling interaction failure (Craveiro et al. 2022 and Selvaraj and Madhavan 2022b). A similar phenomenon was also observed in this paperwork. The test results of the CFS column with pure local failure are compared with design strength classified by a/L_{cr1} ratio in Figure. 3. The assessment indicates that the ultimate load (P_U) of most built-up columns (62.7%) reduced

significantly less than the local failure curve for an a/L_{cr1} ratio of more than 1. The meaning of a/L_{cr1} ratio is given in Selvaraj and Madhavan 2022b and 2023a. The ultimate load reduction should be attributed to the following; (a) the larger plate slenderness leads to a large magnitude of local failure deformation; (b) the shift in the effective centroid induces eccentricity; (c) the eccentricity in loading combined with larger spacing between the fastener connections (a) reduces the individual shape's stiffness (larger a/r_i). Finally, this will lead to local-overall buckling interaction failure. This indicates that the larger intermediate spacing limits suggested by AISI S100 are inadequate to maintain the integrity of the full shape until failure for columns with local failure vulnerability. Further, it should be noted that among the specimens with interaction failure, 89% of them are tested and designed as fixed-fixed-end boundary conditions. This interaction failure in most of the F-F boundary condition columns raises a question on the constraint provided by the support conditions used. The CFS column which has large local slenderness might start failing locally on the plates from near the support ends (not only in the middle half-length). This local failure near the ends reduces the stiffness and leads to a change in its load-displacement plot. This stiffness loss due to end conditions might trigger overall failure. Therefore, the consideration of $K = 0.5$ for fixed-end columns that are subjected to pure local failure ($\lambda_l \gg 1$) may be revised.

3.3 Interaction Failure: Due to Reduction of Shape Stiffness

The research work by Camotim et al. (2020a; 2020b), Dinis et al. (2018) and Young et al. (2018) recommended that the visible interaction failure occurs when a minimum of two F_{cr} values are lesser than F_y . In disparity to this statement, the analysis of the data indicates that the built-up column shapes with local slenderness of more than 1 involve immense interaction failure. This can be observed in Figure. 4a. In addition, it should be noted that all the results presented in Figure. 4a (F_U vs F_{nl}) have an a/L_{cr1} ratio of more than 1 and 62% of them are having fixed-end conditions. Therefore, this interaction failure can be attributed to the reduction of shape stiffness due to combined local buckling deformation caused by a larger a/L_{cr1} ratio leading to the next possible mode of failure prematurely. The detailed investigation of stiffness reduction in a CFS section due to local failure is described in Rasmussen (2023).

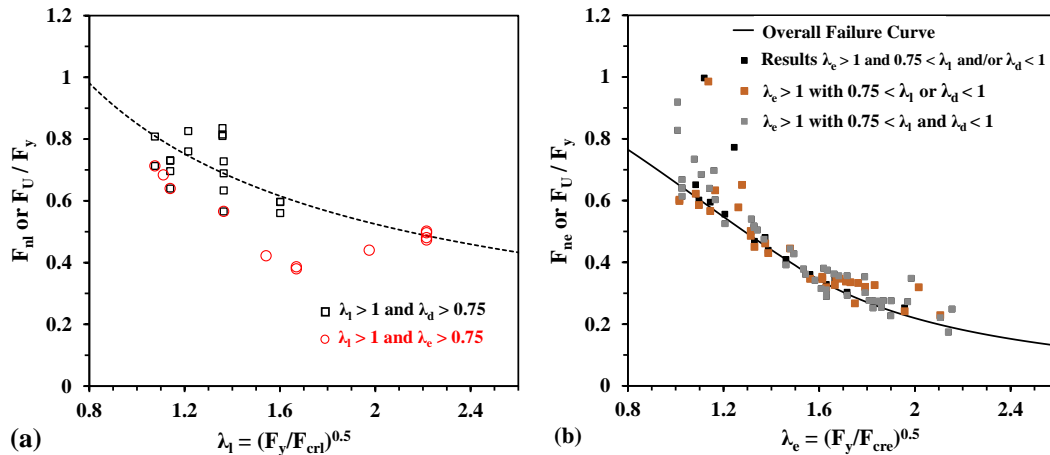


Figure. 4. Interaction failure due to stiffness reduction: (a) Assessment of P_U vs; (b) Assessment of F_U vs. F_{Ne}

The above interaction failure was also observed in columns with overall buckling failure susceptibility ($\lambda_e > 1$). The test results with slenderness ratios $\lambda_e > 1$ are compared with the overall buckling design calculations (F_{ne}/F_y) in Figure. 4b. Points to be noted are: (a) all the columns are

of $a/L_{cr1} > 1$ which triggered plate instability; (b) all the columns are long, and (c) 75% of the unconservative calculations are from fixed-end conditions. Therefore, changing the K value for fixed-end columns may lead to accurate design calculations.

4. Modification in Direct Strength Method Based Design Approach

Most of the CFS built-up shapes subjected to pure local failure are experiencing interaction failure due to built-up assembly shape, spacing between the fastener connections, buckling half-wavelength, shape stiffness and mostly boundary conditions effect ($K = 0.5$) as demonstrated in Figures 3 and 4. Hence, it is advised to use the local-overall buckling interaction curve (F_{nle}) with $K = 0.65$ for fixed-end columns for shapes vulnerable to pure local failure. In addition, it is endorsed to use the modified local slenderness (λ_{lem}) to account for the spacing between the fastener connections effect (a/L_{cr1}). The improved DSM-based design equation for locally slender CFS cross sections is shown here

$$F_{nem} = \begin{cases} F_{nem} & \lambda_{lem} \leq 0.776 \\ \left[1 - 0.15 \left(\frac{1}{\lambda_{lem}^2}\right)^{0.4}\right] \left(\frac{1}{\lambda_{lem}^2}\right)^{0.4} F_{nem} & \lambda_{lem} > 0.776 \end{cases} \quad (1)$$

$$\lambda_{lem} = \sqrt{\frac{F_{nem}}{F_{cr1}} \left(\frac{a}{L_{cr1}}\right)^{0.2}} \quad \text{when } F_{cr1} < F_y \quad (2)$$

Where F_{nlem} (stress) should be multiplied with Area to determine the P_{nlem} (load). The assessment between design calculations from modified DSM and test results is shown in Figure 5 (compare these results with Figures 3 and 4). The assessment indicates that the design load calculations with modified DSM are unconservative only for six CFS built-up shapes that have unstiffened edges and a slenderness range above 1.25. A more detailed discussion about this assessment is available in Selvaraj and Madhavan (2024 - upcoming paper). The reliability indices of the modified approach also exceed the LRFD target reliability index (2.5), which means the proposed DSM-based design approach applies to practice. A considerable number of CFS shapes vulnerable to overall buckling failure are undergoing interaction failure mainly due to the support condition effect (with $K = 0.5$) and may also be due larger a/L_{cr1} ratio effect as illustrated in Figure. 4. This interaction failure can also be counteracted by; (a) using an overall buckling-local and overall buckling-distortional interactive curves; (b) modifying the overall buckling slenderness with a/L_{cr1} ratio; (c) changing the effective length factor. Though all the methods are possible, they will be too conservative. Therefore, it is suitable to modify the overall buckling failure curve to account for this interaction failure (Zhang and Young 2018; Li and Young 2022b). The modified equation for overall buckling is shown here

$$F_{nem} = \begin{cases} 0.9 (0.658 \lambda_c^2) F_y & \lambda_c \leq 1.5 \\ 0.9 \left(\frac{0.877}{\lambda_c^2}\right) F_y & \lambda_c > 1.5 \end{cases} \quad \text{with } \lambda_e = \sqrt{F_y/F_{cre}} \quad (3)$$

The assessment of the modified overall buckling failure curve ($P_{nem} = F_{nem} \times \text{Area}$) is shown in Figure. 5, and it indicates that the modified design curve is conservative. The reliability analysis indicates that the modified overall buckling failure curve calculations are reliable (β_1 and β_2 are more than β_{target}), which means practically applicable. The interaction failure in shape is susceptible to distortional failure due to less post-buckling strength. Therefore, it is appropriate to consider the distortional-overall buckling interaction curve (F_{nde}) to counteract the variation in interaction failure as shown here.

$$F_{ndem} = \begin{cases} F_{nem} & \lambda_{de} \leq 0.561 \\ \left[1 - 0.25 \left(\frac{F_{crd}}{F_{nem}}\right)^{0.6}\right] \left(\frac{F_{crd}}{F_{nem}}\right)^{0.6} F_{nem} & \lambda_{de} > 0.561 \end{cases} \quad (4)$$

$$\lambda_{de} = \sqrt{\frac{F_{nem}}{F_{crd}}} \quad (5)$$

Where F_{ndem} (stress) should be multiplied by the area of the shape to determine the P_{ndem} (load). The assessment between the modified design curves (F_{nem} , F_{nlem} and F_{nde}) and test results are shown in Figure. 5, and it indicates that the modified design curves are conservative. The reliability analysis also indicates that the modified overall buckling failure curve calculations are reliable (β_1 and β_2 are more than β_{target}), which means practically applicable.

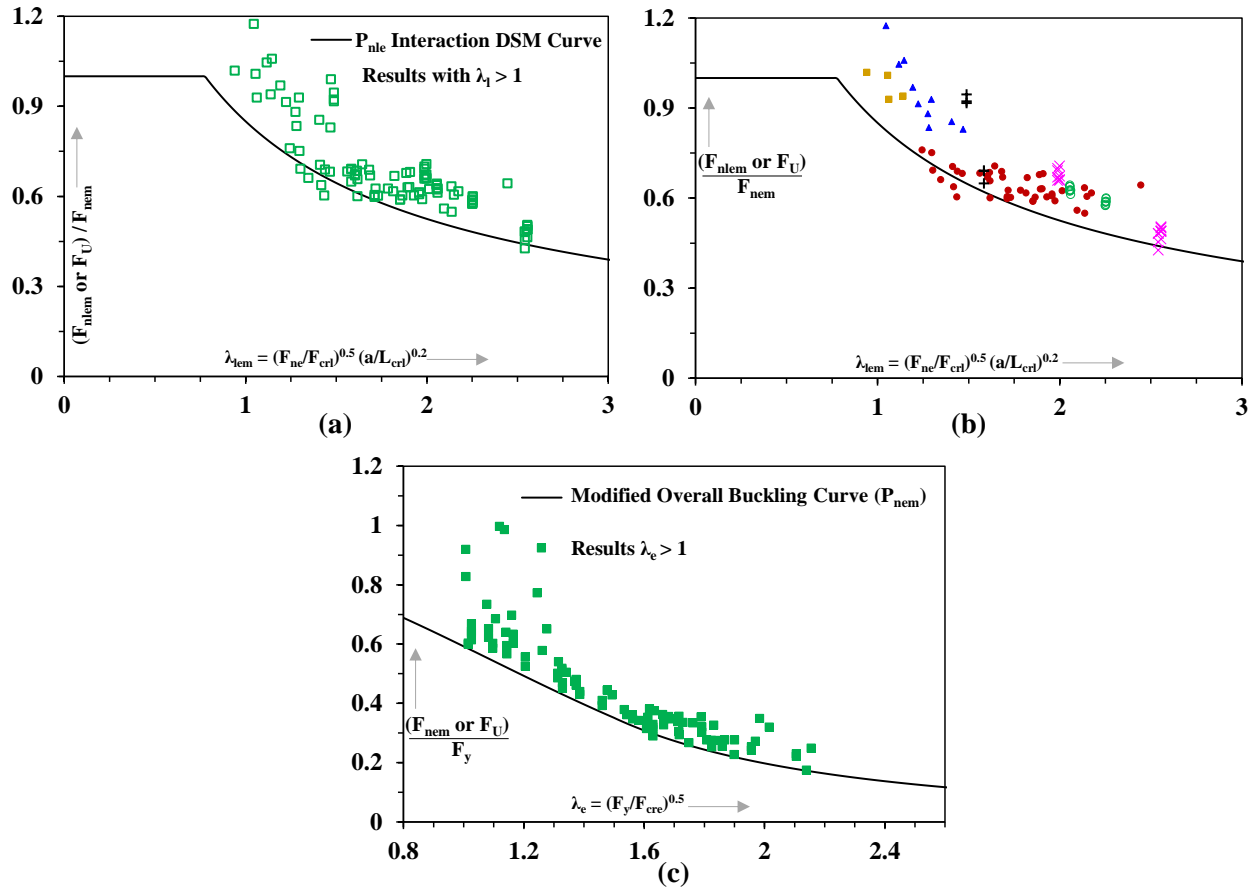


Figure. 5. Proposed DSM curves: (a-b) Modified local failure curve (F_{nle} with λ_{lem}); (c) Modified overall buckling failure curve

Conclusions

The ultimate load of the CFS built-up assembly structural members is affected by interaction failure depending on the shape geometry, end support conditions, P_y/P_{cr} ratio, and $P_{cr} \max/P_{cr} \min$ ratio. This study explores the different factors causing interaction failure in CFS built-up columns that are vulnerable to distinct failure modes from a total of 813 test data collected from the literature. Based on the assessment the following conclusions can be drawn:

- The effect of spacing between the fastener connections in ultimate load reduction should be incorporated in the AISI design specification.

- The assumption of fixed-fixed boundary conditions eroded the ultimate load of the column as one of the main factors, particularly in shapes that have susceptibility to local failure. Therefore, the effective length factor for CFS fixed-end columns subjected to local failure should be modified as suggested.
- Built-up shapes should not be designed with unstiffened elements and a large slenderness range to avoid non-uniform element failure that leads to a shift in the center of gravity.
- A modified overall buckling failure curve is proposed due to interaction failures, the same to be used with other buckling interaction curves.
- The local failure curve should be revised to a local-overall buckling interaction curve with modified local slenderness (similar to Selvaraj and Madhavan 2022b and 2023a) to account for the interaction failure. The effective length factors should also be modified.

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